# Chapter 6
# WELDING

# 6. WELDING

WELDING is the process by which two or more parts are joined by localised fusion of the metal to form a single component; the original contours of the initial parts disappear after assembly.

Arc welding is still by far the most widespread joining process and the one used most frequently in shipbuilding.

Technical advances in arc welding with pulsed MIG have helped improve the performance of welding machines and the quality of the weldments produced.

The development of other processes such as laser beam welding or friction stir welding (FSW) will further advance the design and fabrication of aluminium sub-assemblies for shipbuilding.

Whatever the welding technique that is used, the quality of workmanship of aluminium alloy weldments becomes increasingly important on very long ships as it determines the fatigue resistance of the most stressed areas of the vessel. The weld is a vital element in the fatigue strength of an assembly.

## 1. HISTORICAL REVIEW

The first attempts at welding aluminium were made in 1904, and gas welding was used at that time [1].

Up until the early Sixties, welding with the oxy-acetylene torch was the only method available for welding aluminium alloys. The use of this process was limited to flat welding and thin sheet.

For many years the presence of a natural oxide film on the surface of aluminium was a major obstacle to the welding of this metal. For aluminium to be welded correctly, this film must be removed and prevented from re-forming by shielding the weld pool from the surrounding atmosphere.

In oxy-acetylene welding, fluxes in the form of paste diluted in water were deposited on the edges to be welded and on the filler wire to eliminate the oxide film. These fluxes were based on chlorides and fluorides. To avoid any risk of corrosion from flux residues, these had to be removed by brushing or washing in water.

As with the arc welding of steel, rods of filler metal coated in flux for welding thicker products began to become available from 1925 onwards. One of the very first known applications of arc welding using coated rods came in France in 1934 with the construction of railway vehicles in 5056 (A-G5) alloy for the 'Cie Française des Chemins de Fer du Nord' [2]. This process was not developed to any great extent owing to the unsatisfactory quality of the weldments.

The first attempts at arc welding in a shielding gas (argon or helium) were made in the middle of the Nineteen Thirties [3]. This technique represented a major step forward, and eliminated the need for flux with its attendant risks of corrosion. It was now possible to weld at high speed and in all positions, making aluminium a "fully fledged" fabrication metal in its own right.

The industrial development of the TIG and MIG processes began in the early Fifties and advances and improvements in these processes have been made ever since. One such innovation came at the beginning of the Nineties with electronically controlled pulsed MIG welding.

WELDING A HATCHWAY FRAME

Up until the early Sixties, ships made from aluminium alloys – as well as aluminium alloy equipment on steel ships (superstructures, funnels etc.) – were assembled by means of riveting, as indeed steel ships still were.

The sailing ship "Morag Mhor", a 70 foot ketch made from aluminium-magnesium alloy (4 and 5% magnesium) and designed by a British naval architect, was the first known boat to be constructed with MIG welding in 1953 [4].

# 2. SPECIFICS OF WELDING ALUMINIUM

Although the techniques used for welding semi-finished products made from aluminium alloys are very similar to and even the same as those used for carbon steel, the operating conditions are rather different. This is due to the presence of the oxide film ($Al_2O_3$) on the surface of the metal (1) and to the physical properties of aluminium alloys which are very different from those of steels (table 47, p. 87).

*(1) Cf. Chapter 10.*

*(2) In arc welding with the continuous TIG process, the welded piece is always connected to the minus pole (–) to remove the oxide film.*

*(3) Although it is an electrical insulator, the film is too thin to stop the flow of electrical current in the same way as layers of anodising whose thickness is commonly 15 to 20 microns.*

*(4) Greases and lubricants are carbon chains with the general formula $C_nH_{2n}O$.*

## 2.1 The oxide film

The natural film of oxide which permanently covers the surface of the metal is 50 to 100 nanometers thick. Its melting point is very high, 2052°C, and it is insoluble in solid or liquid aluminium.

For welding purposes the film must be removed (2) and prevented from re-forming while the filler metal is being applied to the weld seam (3). This is why aluminium must be arc welded or laser welded in a controlled atmosphere consisting of an inert gas such as argon, helium or their mixtures.

Although the film is chemically stable (it is an oxide) it nevertheless reacts with its environment by adsorbing traces of rolling mill oils, shaping lubricants and the moisture present in the air. All of these elements are sources of hydrogen (4) when they are dissociated in the plasma of the electric arc.

LNG CARRIERS

## 2.2 Solubility of hydrogen in the fused metal

Given the very high solubility of hydrogen in liquid aluminium, it dissolves in the weld pool of the weld seam as it is formed (figure 60) (5).

However since hydrogen is not soluble in solid aluminium, if cooling is too fast it will have a tendency to become trapped in the metal forming bubbles that will be porosities in the weld seam (6).

This is why it is so important to remove all possible sources of hydrogen on the metal that are present in moisture, in traces of grease and in the shielding gases.

## 2.3 Physical properties

Table 47 presents a comparative list of those physical properties of aluminium and steel which affect the welding of these metals. It is the thermal properties which account for the significant differences between the welding conditions for aluminium compared with those for steel.

Aluminium has a high calorific capacity (899 $J.kg^{-1}.K^{-1}$, compared with 420 for steel) and a higher thermal conductivity (229 $W.m^{-1}.K^{-1}$, against 54 for steel). This means that much of the energy input from the arc is used to heat up the pieces that are to be welded.

Aluminium's high effusivity (7) requires a very high level of welding energy. All other things being equal, the rise in temperature of aluminium parts to be welded will be greater than for steel parts.

Heat quickly dissipates in aluminium due to its high diffusivity (8) (0.9 compared with 0.2 for steel), and this must be compensated by the input of heat from the electric arc.

Aluminium's high coefficient of expansion ($23.10^{-6}K^{-1}$), its high diffusivity and the metal's high level of temperature mean that welding is accompanied by more strain in aluminium than in other metals.

To maintain stable conditions therefore, aluminium must be welded at a rate higher than the rate at which the heat is propagated (figure 61).

If carbon steels are cooled too quickly they undergo a martensitic transformation that is accompanied by an increase in volume which in turn can cause cracks at the base of the weld seam.

**SOLUBILITY OF HYDROGEN**

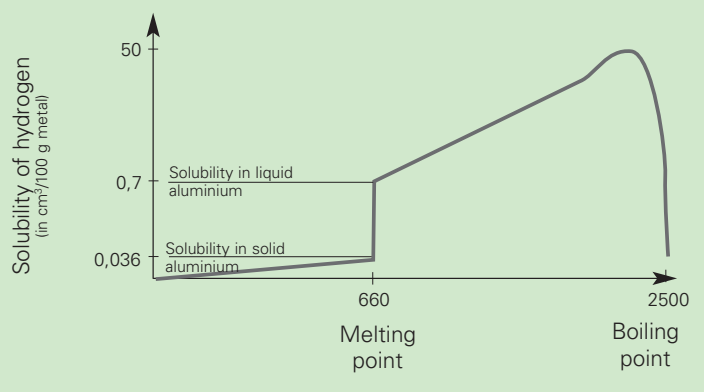



Figure 60

*(5) Contrary to what happens with certain steels, hydrogen does not embrittle aluminium and does not sensitise it to stress corrosion.*

*(6) Cf. table 54, pp. 104-105.*

*(7) Effusivity "b" is the product of the square root of the product of thermal conductivity λ by density ρ and by specific heat capacity Cp:*
$b = \sqrt{\lambda.\rho.Cp}$.

*This variable describes the amount of heat which a heated zone receives by conduction, where:*
*λ = thermal conductivity*
*ρ = density*
*Cp = calorific capacity.*

*(8) Thermal diffusivity "a" is defined by the relation*
$a = \frac{\lambda}{\rho.\ Cp}$

**HEAT FLOW DURING WELDING**

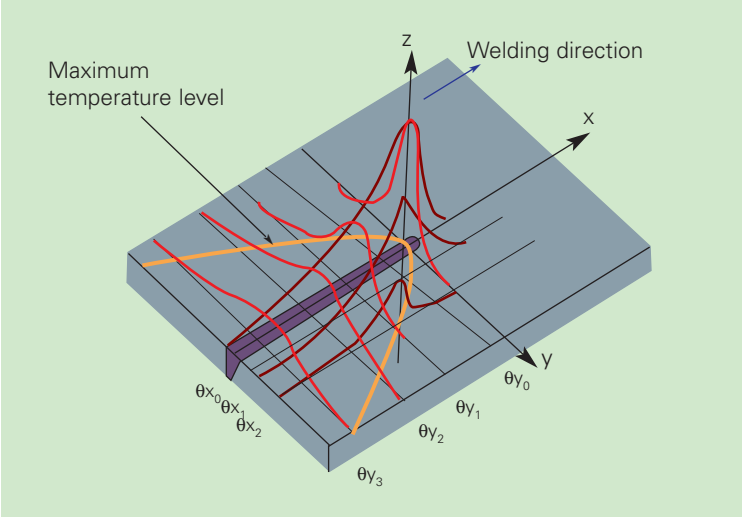



Figure 61

There are no such changes with aluminium alloys. As a result, the rapid cooling rate of the weld – between 100 and 1000°C per second – does not cause any defect in the seam.

It is therefore not normally necessary to preheat aluminium prior to welding as can be done with steel (to prevent cracks in the weldment during cooling).

All of these factors must be allowed for when designing joints and executing the actual welds. This highly important aspect is discussed in Section 3.

*(9) The mechanical properties of the weld seam itself are usually superior to those of the HAZ. When there is a crack, this usually occurs outside the weld seam.*

## 2.4 The heat affected zone (HAZ)

With steels, changes of phase cause local hardening of the heat affected zone over a width of several millimetres. In contrast, the effect of heating is to soften aluminium alloys when they are:

- in the strain hardened condition as is the case with the 5000 series alloys in the H116, H24, H32 and H34 tempers,
- thermally treated, e.g. 6000 series alloys in the T5 or T6 tempers.

As figure 62, p. 88, shows, the material's mechanical properties change gradually from the weld seam outward to the edges of the HAZ (9).

As a result, the mechanical properties of the weldment are usually inferior to those of the parent metal. They are near to the annealed condition for strain hardened alloys and to the T4 temper for age hardening alloys.

**PHYSICAL PROPERTIES OF ALUMINIUM AND STEEL E24**

| Property | Unalloyed Aluminium (1050A) | Steel E24 | Ratio Al/Steel |
|---|---|---|---|
| Fusion interval (°C) | 645/658 | 1 400/1 530 | |
| Melting point of oxides (°C) | 2 052 | 900 | |
| Solidification shrinkage (%) | 1,7 | 1,2 | 1,4 |
| Density ($kg.m^{-3}$) | 2 700 | 7 820 | 0,34 |
| Mass heat capacity Cp ($J.kg^{-1}.K^{-1}$) | 899 | 420 | 2,14 |
| Latent heat of fusion ($kJ.kg^{-1}$) | 385 | 210 | 1,83 |
| Thermal conductivity ($W.m^{-1}.K^{-1}$) | 229 | 54 | 4,24 |
| Thermal diffusivity "a" ($10^{-4}.m^2.sec^{-1}$) | 0,9 | 0,2 | 4,5 |
| Effusivity "b" ($J.\ m^{-2}.K^{-1}.sec^{-1/2}$) | 24 000 | 16 000 | 1,5 |
| Coefficient of linear expansion $\alpha$ ($10^{-6}.K^{-1}$) | 23,5 | 13,5 | 1,74 |
| Electrical resistivity $\rho$ ($10^{-9}\ \Omega.m$) | 292 | 1010 | 0,29 |
| Young's modulus (MPa) | 70 000 | 210 000 | 0,33 |

Table 47

This is why for stress calculations, the design codes and regulations of the classification societies specify levels of yield strength of the annealed condition for strain hardening alloys and of the T4 temper for age hardening alloys.

In the case of butt welds, the HAZ is some 25 mm wide either side of the weld seam whatever the thickness of the parent metal and whether TIG or MIG welding is used.

The impact on the mechanical properties of the weldment must be minimised by:

- selecting the filler metal that is best suited to the parent alloy,
- establishing a welding cycle that is as rapid as possible to limit the size of the HAZ,
- designing the assembly so that the weldments are in the least stressed positions.

## 2.5 Weldable aluminium alloys

Most of the alloys belonging to the 1000, 3000, 5000 and 6000 series can be welded with the conventional TIG or MIG processes and by the other energy beam processes – electron beam, laser etc.

The 5000 series is the one that is most suitable for welding.

The wrought alloys *containing copper* in the 2000 and 7000 series (10) are not easy to arc weld (11). The presence of the copper causes cracking and shrinkage cracks on the weld seam.

The weldability of castings depends more on the method of casting than on the chemical composition of the alloy. Diecast parts cannot be welded as they may contain a lot of air that is introduced during the casting process.

Sand cast or chill cast parts can be welded provided they are "clean", i.e. free from porosities and shrink holes that produce blisters in the weld seam (12).

The 42100 (A-S7G) and 43300 (A-S10G) alloys can be welded with alloys in the 6000 series (13).

*(10) The 7000 alloys without copper such as the 7020 can be welded but the HAZ is very sensitive to exfoliating corrosion (cf. Chapter 10).*

*(11) The welding regulations and construction codes such as Eurocode 9 and standard BS 8118 actually exclude these alloys from arc welded assemblies.*

*(12) Porosities in sand castings can be avoided by the use of metal heat sinks that improve the microstructure and prevent the formation of porosity.<0}*

*(13) The 5000 series cannot be welded with the 42000.*

**MECHANICAL CHARACTERISTICS OF THE HAZ**

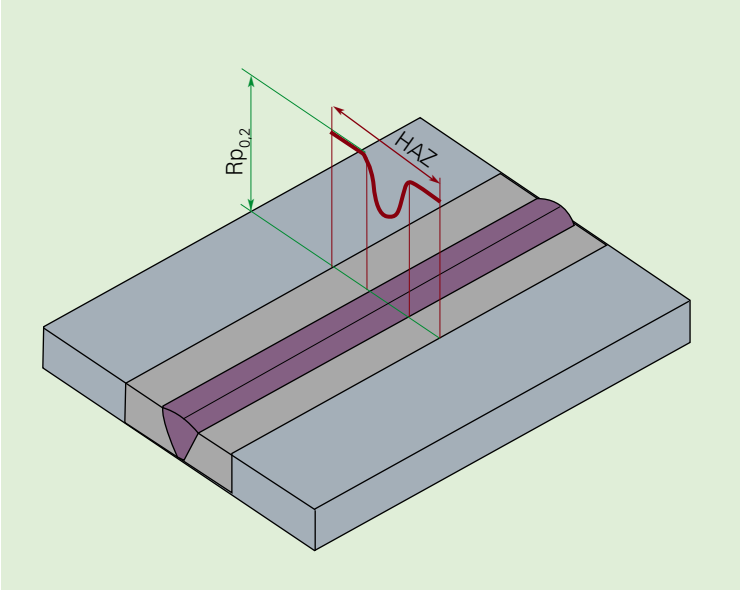



Figure 62

# 3. IMPLICATIONS FOR THE DESIGN AND EXECUTION OF WELDS

During welding, every point on a welded piece undergoes a heat cycle whose profile is a function of a number of parameters:

- the power of the heat source which depends on the process (MIG, TIG etc.),
- the geometry of the piece,
- the welding position (flat, vertical, horizontal, overhead),
- the diffusion coefficient of the material.

The result is a variety of different temperatures present in the component in the course of welding. These temperature differences translate into residual distortions of varying extent and which are due:

- to differences in expansion,
- to shrinkage as the weld seam solidifies.

## 3.1 Distortion

Distortion can be longitudinal or transverse:

- **longitudinal distortion** is caused by the contraction of the metal during the process of cooling which is not uniform. Stresses are set up along the weld seam (figure 63).

These stresses depend on the position of the weld – they are minimal or non-existent when the weld is on or near the neutral axis of the piece (figure 64) but very pronounced when the weld is asymmetrical. Concavity will follow the same orientation as the weld seams. On long components, distortion may manifest itself as a twist that will prove difficult to correct (figure 65).

As a general rule, distortion will be significant when the weld seams are asymmetrical relative to the piece.

- **transverse distortion** is due to a shortening of the weld seam – this is more pronounced at the surface of the seam than at its root and so creates a 'gripping' effect with angular deformation (figure 66).

This effect must be limited by balancing the stresses with a second weld: a double vee-groove weld on thick components and on the opposite side for fillet welds (figure 67).

LONGITUDINAL DISTORTION

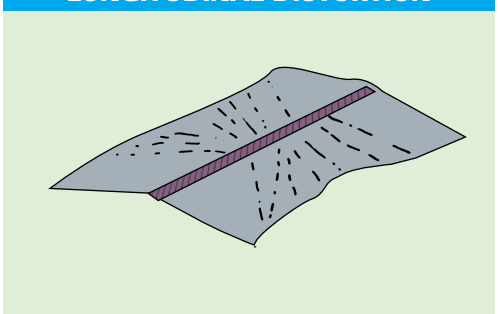


Figure 63

WELDS ON THE NEUTRAL AXIS

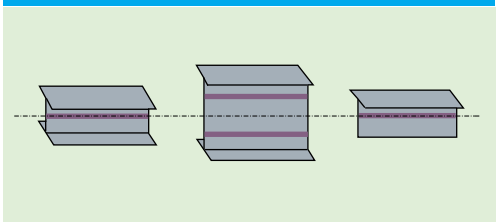


Figure 64

ASYMMETRICAL WELDS

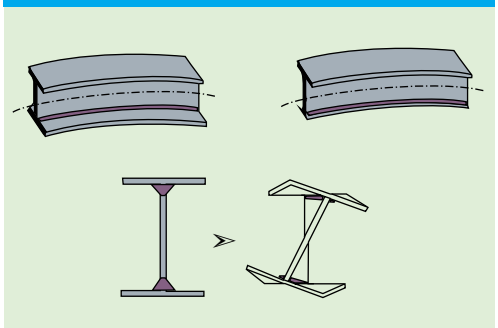


Figure 65

DISTORTION OF WELDS

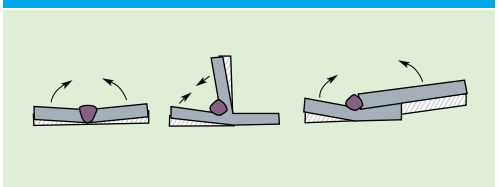


Figure 66

CORRECTION OF DISTORTION

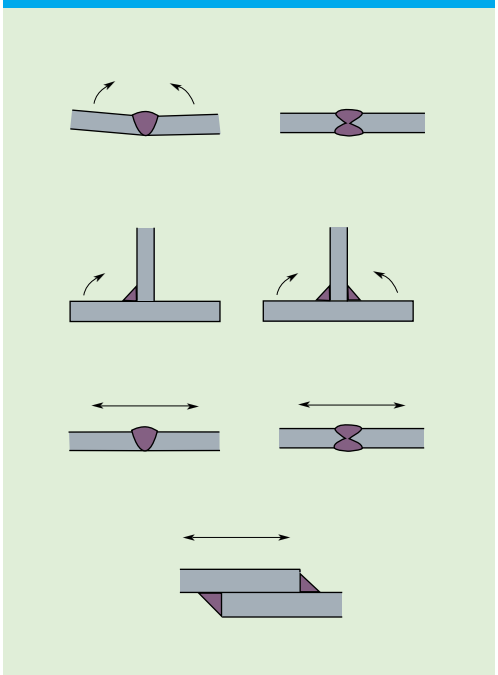


Figure 67

## 3.2 Stresses

Local levels of stress can exceed the yield strength of the metal, and are a function of:

- the shape of the piece,
- the layout of the weld seams,
- the weld sequence,
- positioning tools,
- clamping.

Figure 68 indicates the level of longitudinal residual stress in sections fabricated by welding.

## 3.3 Controlling distortion

There are a number of factors by which distortion can be controlled:

- **Joint configuration**: so far as possible the designer should position the welds in a plane of symmetry of the component and on its neutral axes (14), and use specially designed shapes (15) where possible.

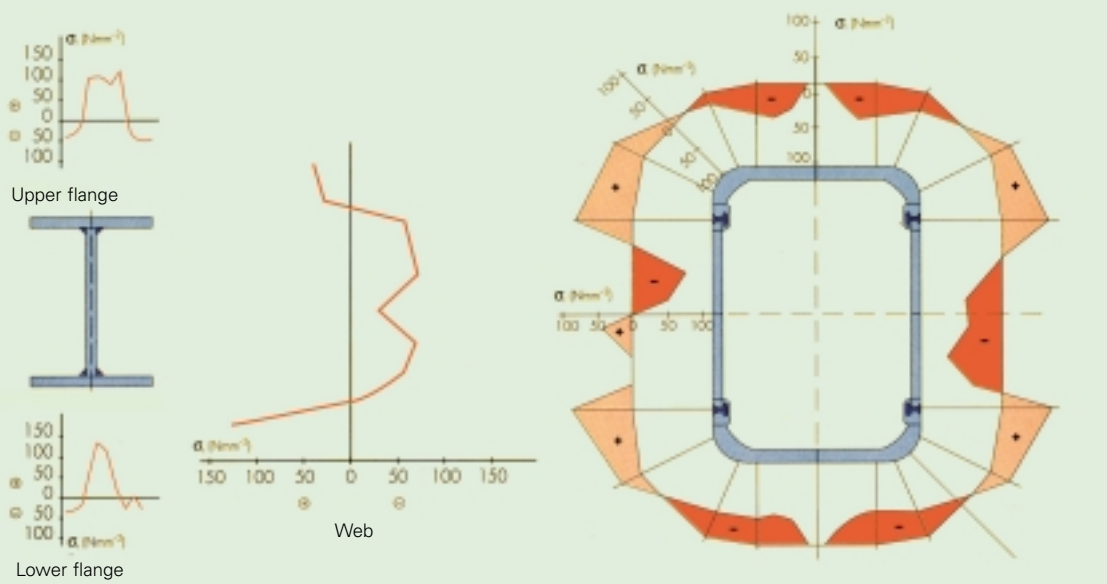
- **Careful forming** must be used to minimise clearances and offsets between the welded components and to eliminate fitup errors,

- **The welding sequence**: it is important to weld moving towards the outside edges to allow expansion to occur freely. The welds must be executed in reverse order of length, with the shortest first to better distribute any distortion. Distortion can be corrected more easily with long seams (figure 69).

Where possible, automatic welding with two torches is a very good way of reducing distortion (figure 70).

The distribution of internal stresses can be optimised by using a sequence of welds that

**RESIDUAL STRESSES**




Connection of plates in 5083 H111 and extruded shapes in 6082 T6

Figure 68

YACHT FRAME

**WELD SEQUENCE**

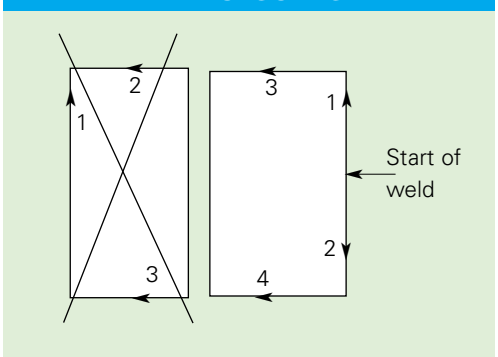



Figure 69

induces residual compression stresses in the weld seams that are stressed in tension (figure 71). This approach will very significantly improve the fatigue strength of the welded joints.

■ The purpose of **clamping** is to hold the parts in position. However clamps, fixtures etc. must not prevent expansion on the longest axes or components will be stressed (with an attendant risk of softening due to expansion Δl). Generally speaking, parts should not be clamped in the directions of greatest expansion as this will aggravate distortion square to the weld (figure 72).

On thick material, angular distortion can be avoided by attaching temporary stiffeners across the weld (figure 73).

■ **Welding parameters**: Distortion can also be caused by the solidification shrinkage of the weld seam. The greater the quantity of fused metal and the higher the temperature, the greater this shrinkage will be. Distortion can be limited by using high energy heat sources to weld as rapidly as possible – the faster the welding, the less time the heat has to dissipate

■ **Balance of heat flows:** Longitudinal distortion can be aggravated by a thermal imbalance between the welded pieces. This imbalance can be due to:
- the very different masses of the pieces that are to be joined,
- misalignment as the welding torch moves,
- poor contact with the support,
- etc.

Whenever possible therefore, the "thermal masses" must be balanced to achieve good thermal symmetry in the assembly (figure 74).

*(14) This layout has a positive influence on the fatigue behaviour of the welded assembly, cf. Chapter 4.*

*(15) Cf. Figures 11 to 14, Chapter 2.*

**WELDING WITH TWO TORCHES**

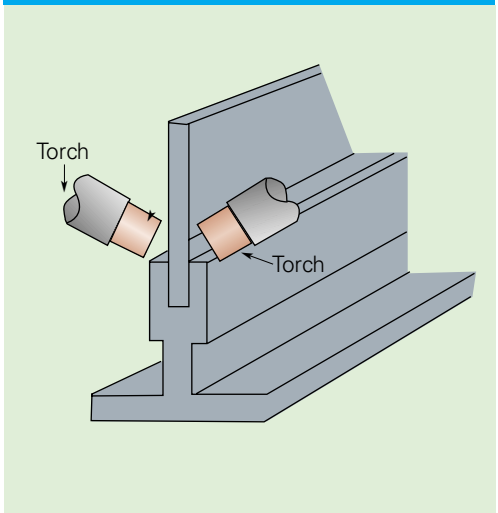



Figure 70

**BUTT WELDING SHAPES**

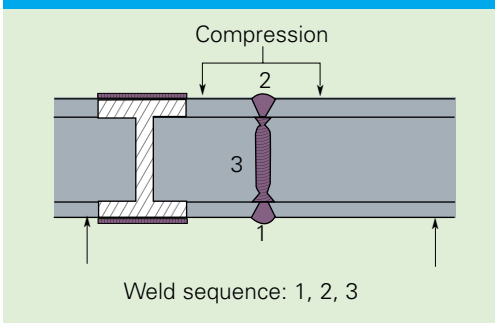



Figure 71

**EFFECT OF CLAMPING**

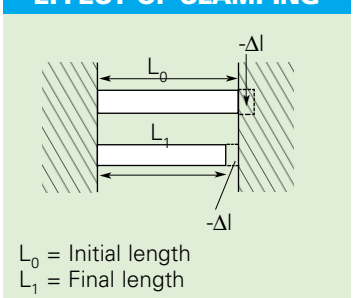



Figure 72

**THERMAL BALANCE OF A WELD**

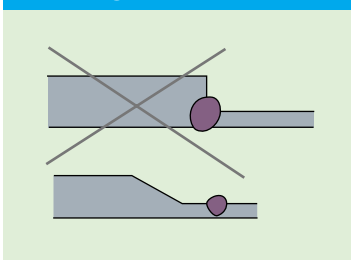


Figure 74

**CLAMPING WITH STIFFENERS ACROSS THE WELD**

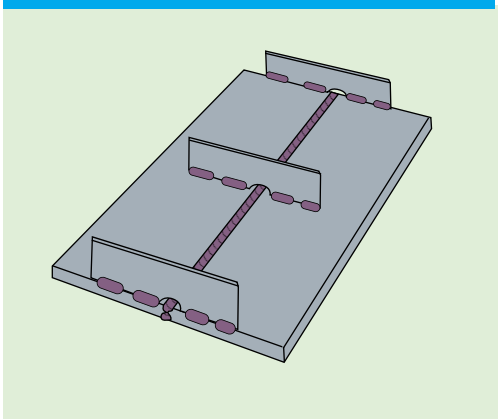


Figure 73

## 3.4 Use in the fabrication of a section with stiffeners

Figure 76 shows the optimum welding sequence (i.e. order in which the welds are made) for limiting distortion when welding the elements of the section illustrated in figure 75.

Welding starts in the centre of the panel and moves out towards the free edges to allow unimpeded expansion.

LIFTING ALUMINIUM SUB-ASSEMBLIES

PANEL WITH STIFFENERS

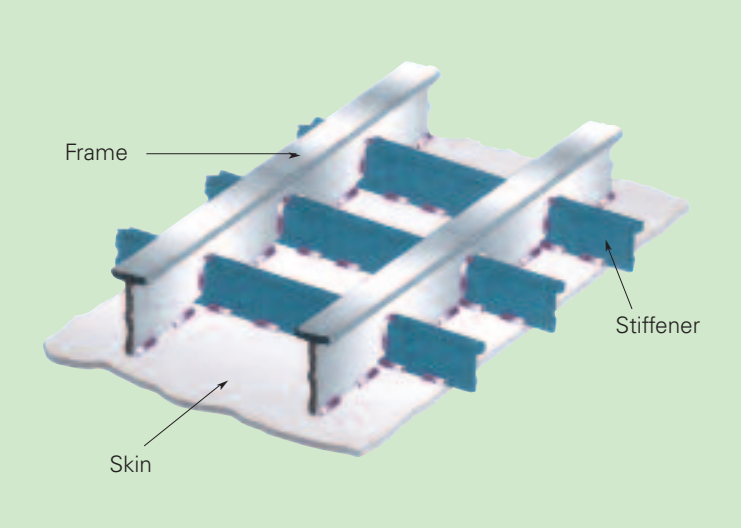



Figure 75

TYPICAL WELDING SEQUENCE OF A PANEL

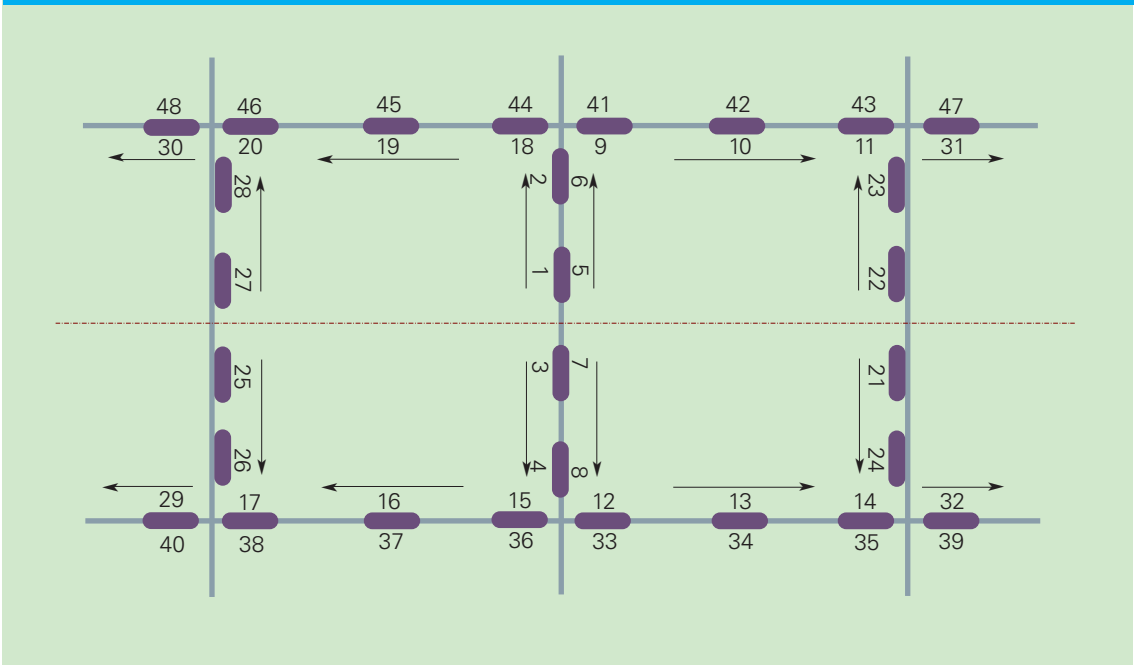



Figure 76

# 4. ARC WELDING PROCESSES

In arc welding, the joint between the pieces to be assembled is made by filling an appropriate shape (Vee, cross, bell) with a filler metal (rod or wire) which is melted step by step. The joint can be filled in one or more passes. As the filler metal melts, so do the edges of the components that are being joined together (unlike in brazing).

Ever since the arc welding of aluminium in inert gases (argon or helium) came into widespread industrial use, there have always been two main processes but they tend to complement rather than compete with one another (table 48, p. 97). One, TIG, is mainly manual, while the other, MIG, can be fully automated. MIG welding has advanced in great strides since the early Nineties to the point where the conditions under which aluminium is welded are now greatly enhanced.

The mechanical properties of the weld seams are identical in both processes, all other things being equal, i.e. parent alloy, filler metal and material thickness.

**PRINCIPLE OF TIG WELDING**

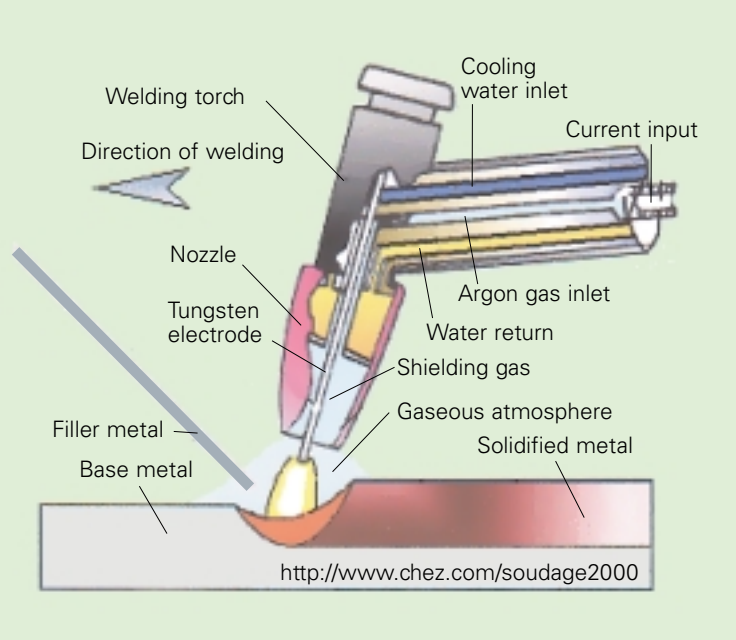



Figure 77

## 4.1 TIG welding (Tungsten Inert Gas)

In TIG welding (16), the electric arc forms between a refractory tungsten electrode and the piece to be welded. The shielding gas – usually argon – is blown out through the nozzle of the torch (figure 77).

In manual TIG welding, the filler metal in the form of a straightened wire rod (0.8 mm to 4.0 mm in diameter) is held manually by the welder. In automated TIG welding, the filler metal is fed automatically from a reel of wire of diameter 0.8 mm to 2.0 mm by a motorised dispenser.

Welding machines operate with stabilised HF alternating current for manual welding or continuous or pulsed d.c. current for automatic welding. Machines must be fitted with an electronic circuit board designed for aluminium welding, with a pulse arc stabiliser and an arc re-igniter.

The geometry of the refractory electrodes is an important factor influencing the quality of the weld. The electrode must be ground sharp unless the welding machine runs on a.c. current. For d.c. current, the electrode tip must be inside a cone of 30 to 60 degrees, and machining (or grinding) marks must run parallel to the longitudinal axis of the electrode (figure 78).

**GRINDING OF TIG ELECTRODES**

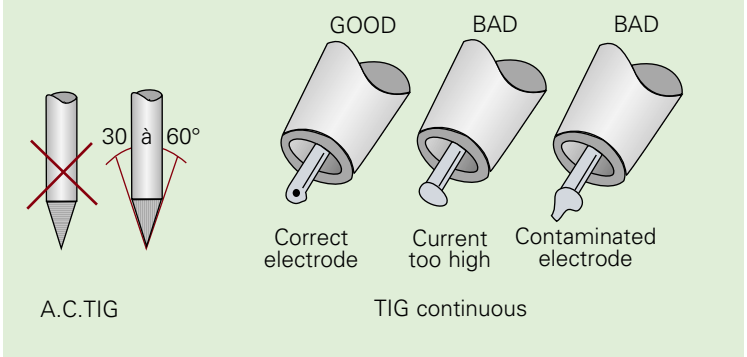



Figure 78

*(16) The process is known as WIG in Germany (tungsten in German is 'Wolfram') and GTAW in America (Gas Tungsten Arc Welding).*

TIG uses less power than MIG, so the heat affected zone is wider (because of the diffusion coefficient) and there is more distortion due to expansion. The rate of welding which is controlled by the welder is relatively slow, in the region of 0.2 $m.min^{-1}$.

TIG welding is above all a manual process and simple to use, allowing meticulous workmanship and precision results.

Welding is possible in all positions. It is suitable for material 1 to 6 mm thick. It can be used to weld with clearances that are over twice the thickness of components under 1.5 mm thick.

TIG is difficult to automate so is limited to use in the development of prototypes and in the repair of defective welds.

## 4.2 MIG welding (Metal Inert Gas)

In MIG welding (17), the filler wire also acts as the electrode supplying the power (figure 79). The wire is automatically uncoiled from a reel and fed to the welding tool (gun or torch) as it is used up.

The welding power is proportional to the amount of wire that is fed to the weld seam, and is supplied by a d.c. power source which can be continuous or pulsed. Connection is made with reverse polarity, i.e. the workpiece is always connected to the minus (negative) pole to ensure descaling of the oxide film.

MIG welding is 'self-pickling' because the transfer of electrons from the workpiece to the filler wire breaks the oxide film (provided it is very thin, several nanometers).

A thick oxide layer that has formed following long exposure to ambient humidity cannot be fully removed, and the weld seam will have oxide inclusions (defect 303, cf table 54, p. 104). Semi-finished products should therefore be stored under cover in a dry place (18).

The welding current varies from 40 to 700 Amps depending on a number of parameters such as the diameter of the filler wire, the position of the weld, the size of the components etc.

The classic MIG process using continuous current has many advantages:

- excellent productivity due to the high rate of filler metal deposition,
- good penetration,
- low splatter,
- the process can be automated.

## 4.3 Synergic pulsed MIG

MIG welding has made great advances since the appearance in the early Eighties of so-called "synergic pulsed current" generators in which the current is supplied by power transistors.

Prior to this, power was supplied by thyristor generators whose pulse frequency was a direct function of the mains frequency. Settings were difficult and lacked flexibility because the speed of the wire had to be adjusted according to the frequency.

Synergic pulsed current generators allow the welding cycle to be regulated (figure 80) to give:

- high current at the start of the weld to avoid lack of fusion and penetration, and
- low current at the end of the weld to prevent crater formation.

**PRINCIPLE OF MIG WELDING**

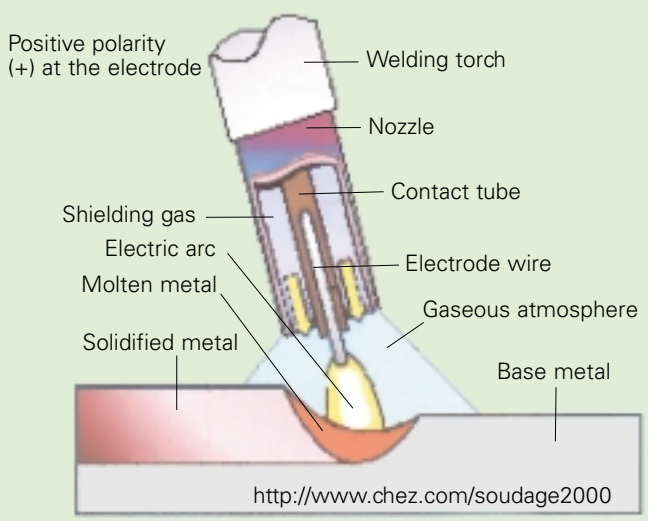



Figure 79

*(17) Still also known as MAG (Metal Active Gas) or GMAW (Gas Metal Arc Welding).*

*(18) Cf. Section 5.*

The welder can control three parameters to optimise the weld seam:

- the speed of the wire, proportional to the welding current,
- the welding speed,
- the height of the arc, proportional to the welding voltage.

With these machines, the parameters adjust automatically to the displayed speed of the wire. Settings can be refined by adjusting the height of the arc.

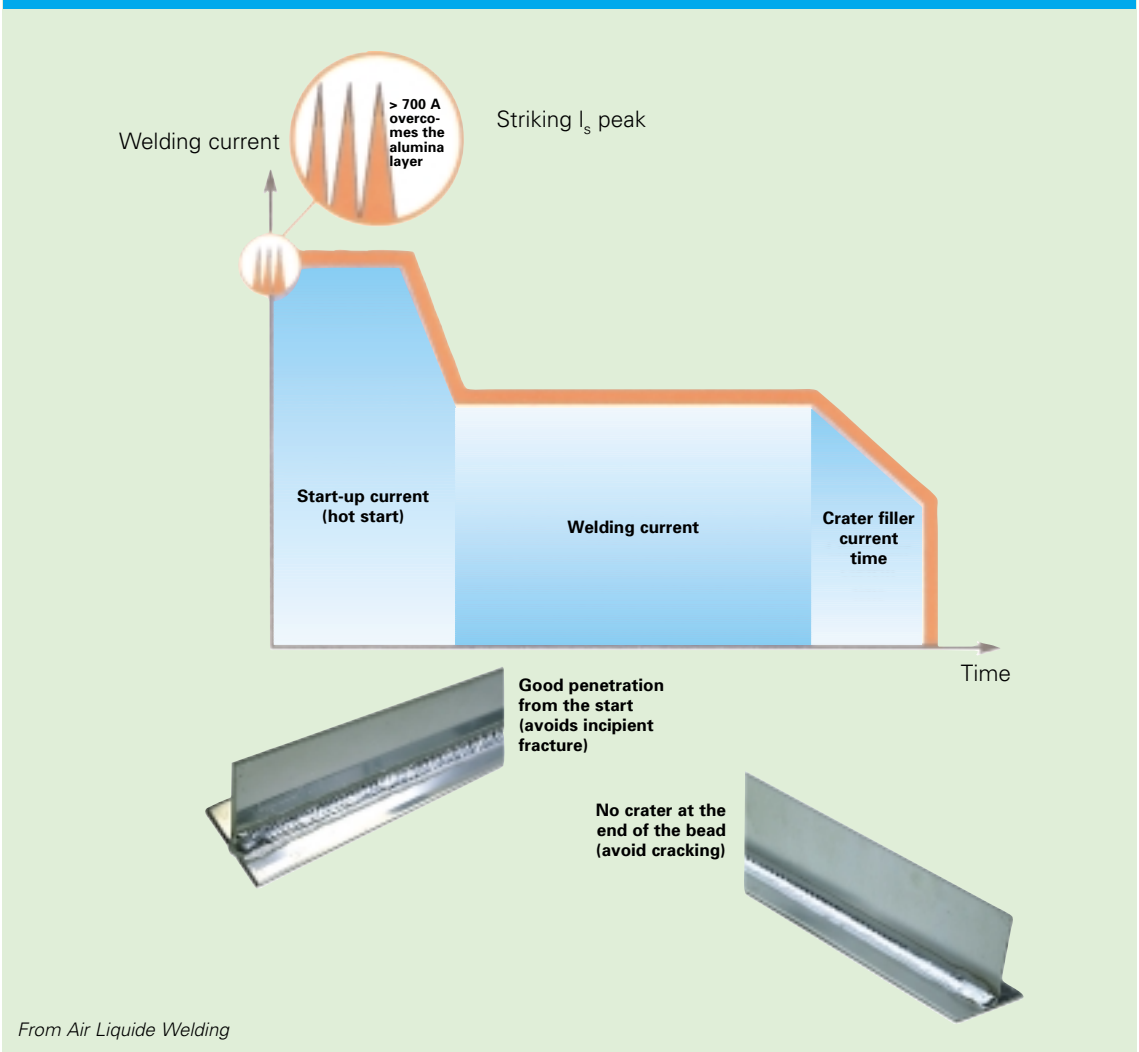
In this system, the metal is transferred "drop by drop" (i.e. one drop of metal per pulse), allowing the minimum weldable thickness to be reduced from 3 to around 1 mm (19).

Pulsed MIG offers a number of additional benefits over conventional MIG welding with continuous current:

- welds can be made in any position,
- distortion is limited (low power input),
- limited weld repairs and fewer in number,
- wide range of thicknesses with the same diameter wire,
- good joint quality and good mechanical properties,
- good appearance of the weld seam, especially with spray transfer,
- process can be fully automated.

*(19) With the old-type generators the transfer of metal by spraying was only possible at 20 V and over. Below this voltage, globule or short-circuit transfer is unsuited to the welding of aluminium, which accounts for the minimum thickness of 3 mm.*

**SYNERGIC WELDING CYCLE**




*From Air Liquide Welding*

Figure 80

## THE SPRAY-MODAL PROCESS

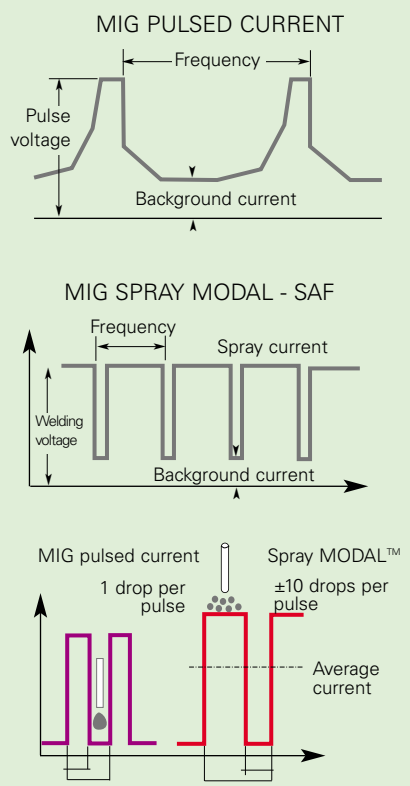



Figure 81

## 4.4 "Spray MODAL" synergic MIG with modulated current

There is now a variant of the synergic welding technique – the "Spray Modal" process (20). It operates with modulated current which falls very rapidly over a very short period of time (several milliseconds) with every pulse during which several drops of filler are projected into the weld pool (figure 81). These rapid variations in voltage within the arc cause the weld pool to vibrate, encouraging the evacuation of hydrogen bubbles from the metal while it is still liquid.

Compared with synergic pulsed MIG, Spray-MODAL welding

- reduces or even eliminates porosity in the weld (figure 82).
- enhances penetration,
- increases welding speed.

## 4.5 Filler wires

An evenly dispensed filler wire will ensure good arc stability and hence the quality of the weld.

The low rigidity of the filler wires requires the use of suitable dispensers to minimise the chances of the wire snagging in the torch tube which must be made of PTFE ("Teflon") to eliminate risks of abrasion.

A torch with a push/pull wire dispensing system is recommended to ensure optimum dispensing regularity, especially when using the 4043A wire grade and in automated welding.

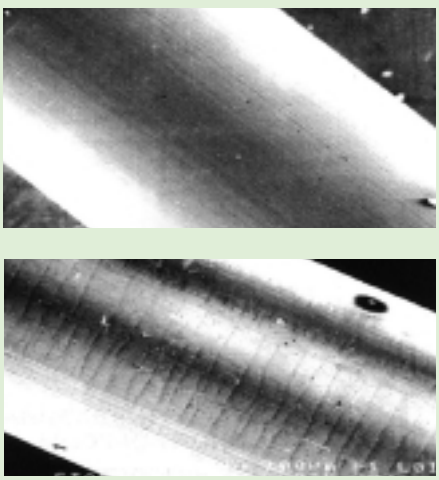
Filler wire is usually 1.2 mm in diameter, although there are also 1.6 mm gauge wires; these are more rigid and their use is growing with pulse MIG. They are also used when the rate of deposition is high.

Shaving the filler wires in the final drawing pass has a number of effects, all of which enhance the quality of the weld:

- it eliminates the outer zone which can be the site of magnesium segregation,
- it removes traces of grease,
- it 'sizes' the weld which removes surface irregularities that are areas of moisture retention (figure 83).

*(20) Patented by Air Liquide.*

## SURFACE CONDITION OF FILLER WIRES

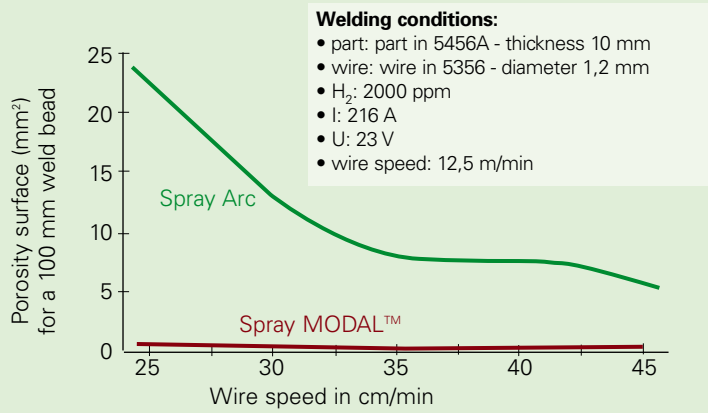


GX50

*From Air liquide Welding*

Figure 83

## EFFECT OF SPRAY-MODAL ON POROSITY




*From Air liquide Welding*

Figure 82

| PARAMETERS OF TIG AND MIG WELDING | | | |
|---|---|---|---|
| | d.c. TIG | a.c. TIG | d.c. MIG |
| Current source | Direct | Alternating with HF and arc decay (specially designed for aluminium alloys). | Direct, with very shallow trailing edge. A pulsed source is a good option for slender work. |
| Electrodes | Zirconium tungsten | Pure tungsten | Filler wire |
| Torch angle | 80° in the direction of advance | 80° in the direction of advance | 80° in the direction of motion |
| Gas | Helium | Argon or mixture of 70% argon, 30% helium(*)<br>Flow 10 $l/min^{-1}$ | Argon or a mixture of 30% argon, 70% helium (*), (**)<br>Flow 1 $l/min^{-1}$ for a nozzle 18 to 25 mm in diameter |
| Welding speed | 0.30 to 0.60 $m/min^{-1}$ | Slow: 0.15 to 0.30 $m/min^{-1}$ | Faster: 0.40 to 1 $m/min^{-1}$ |
| Application | Thickness 0.1 to 10 mm<br>Automated welding with good weld quality | Thickness 1 to 6 mm<br>Prototypes<br>Repairing defective welds | Thickness 1 mm and over, in several passes if necessary<br>All welded fabrications |

*(*) The helium in argon/helium mixtures increases the welding speed and improves penetration.*
*(**) Pulse MIG and Spray MODAL™ synergic MIG methods operate mainly with argon.*

Table 48

# 5. STORAGE OF SEMI-FINISHED PRODUCTS AND FILLER WIRE

Given aluminium's very strong affinity for hydrogen when in the liquid state (figure 60, p. 86), it is essential to remove all possible sources of that element, especially moisture which can deposit on semis and filler wire in storage and hydrate the oxide layer.

Filler wire is always supplied in sealed packs that must be stored in an enclosed, covered room that is at the same temperature as the welding shop. The packs should not be opened until required for use.

When welding operations are complete, any wire left on the reel must be stored in a cabinet maintained at a constant 40°C.

## 6. SURFACE PREPARATION

Other sources of hydrogen are the rolling and forming greases and oils left on the surface of the metal, and other impurities of different types, such as traces of paint.

The surface of the metal must therefore be cleaned very carefully on both sides, starting by degreasing with a non-chlorinated solvent to dissolve the greases and oils (21). Solvents are themselves hydrocarbon compounds containing hydrogen atoms, so great care must be taken to ensure no trace is left prior to welding.

After degreasing, the edges must be brushed (after chamfering as necessary) on both sides of the metal and over a sufficient width that is at least equal to the width of the heat affected zone, i.e. 25 mm. A rotary brush with stainless steel wires should be used for this.

Whatever method of brushing is used (manual or mechanical) the brush itself must be very clean and operators must wear gloves.

The "life" of surface preparation is certainly no more than one day, after which time the oxide film may well absorb moisture once more, especially in humid environments (22).

To eliminate moisture, just prior to welding an oxy-acetylene torch can be used to pre-heat the edges at a temperature above dew point in the region of 30 to 40°C.

## 7. JOINT PREPARATION AND SETUP

These operations are very important, and will determine the quality of the weld and its fatigue resistance. For example, excessive clearance between the workpieces can cause the weld seam to collapse and lead to the formation of undercuts that can be very detrimental to the quality of the weld and its fatigue resistance.

The type of edge preparation will depend on:

- the thickness of the work,
- the type of weld: butt, flat or fillet, vertical, overhead or horizontal,
- the use of a backing strip or bar, whether permanent or not.

As a general rule, the edges of material up to 4 mm thick are not chamfered.

Ideally, edges that are to be welded should be prepared by machining with a coarse-tooth cutter or if this is not available, manually using a coarse file. Avoid grinding with corundum or resin wheels.

Workpiece configuration is also important; this relates to:

- the clearance between the workpieces – this must be as small as possible (23) to prevent distortion,
- the size and shape of the backing bar (stainless steel).

Tables 49 and 50 illustrate a number of examples of edge preparation and configuration found in shipbuilding.

## 8. FILLER METAL

The filler metal must be compatible with the chemical composition of the parent alloys that are to be welded, and must ensure the best possible weldability.

The choice will also depend on the mechanical properties and corrosion resistance that the joint is required to have.

For the aluminium alloys that are used in shipbuilding (and other marine applications), the filler metals are:

- silicon alloys, mainly 4043A, 4045, 4047A,
- magnesium alloys, mainly 5356, 5183, 5556A.

Their compositions are shown in table 51, p. 100.

Table 52, p. 101 - taken from EN 1011–4 (24) - shows possible choices of filler metal according to the hierarchy of criteria used for the weld. 5183 is the best filler metal for welding *Sealium*®.

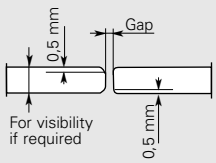
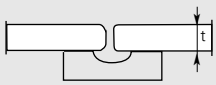
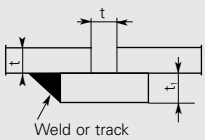
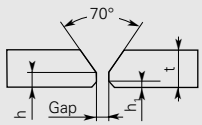
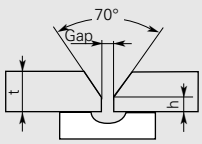
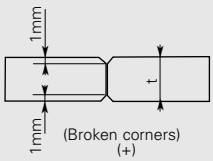
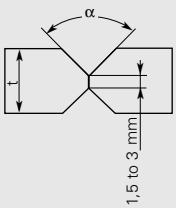
*(21) Chemical pickling in alkaline baths should be avoided at all cost. Thorough washing is essential and experience shows that this is often inadequate, with a risk of subsequent corrosion by traces of the alkaline medium.*

*(22) BS 8118 "Structural use of aluminium, Part 2 Specification for materials, workmanship and protection" states that the time between cleaning and welding must not exceed 6 hours.*

*(23) Zero clearance is the ideal.*

*(24) Standard EN 1011-4. Welding – Recommendations for welding of metallic materials. Part 4: Arc welding of aluminium and aluminium alloys.*

## EXAMPLES OF EDGE PREPARATION FOR BUTT WELDING MIG WELDING

| Position | Welding | Backing | Thickness (mm) | Preparation | Remarks |
|---|---|---|---|---|---|
| All | 1 side only | none | $3 < t \leq 6$ |  | Max. gap 1.5 mm<br>Back-weld advisable for t > 4 mm (*) |
| | | temporary | |  | Max. gap 3 mm |
| | | permanent | |  | $t_1$ = t + 1 mm<br>with max. 6 mm |
| Flat, vertical, overhead | 1 side only | none | $3 < t < 25$ |  | Max. gap 1.5 mm<br>Back-weld advisable for h = 3 mm (*) |
| | | temporary | |  | Max. gap 2 mm |
| Flat | 2 sides alternately | none | $8 < t < 30$ |  | Material over 12 mm thick should be welded automatically with a high current (+). Improvement and visibility of the weld |
| Flat, vertical, overhead | 2 sides alternately | none | $t > 10$ |  | α = 70/90° for flat and overhead welds<br>α = 70° for vertical welds |

*(*) Where a back-weld is advisable, it must be welded after gouging to the base of the first pass.*

*(*) Taken from standard NF 87-010* "Aluminium et alliages d'aluminium – Soudage – Préparation des bords" (Aluminium and aluminium alloys – Welding – Edge preparation).

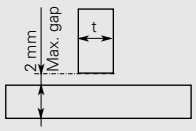
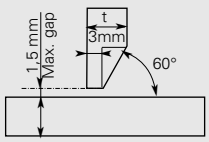
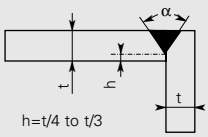
Table 49

# 9. FINISHING

The purpose of weld finishing operations is to:

- repair defective weldments,
- remove any black deposits left by welding,
- correct structures with excessive distortion,
- shave the seam,
- put the seam in compression by shot-peening,
- complete the concavity of the seam.

## EXAMPLES OF EDGE PREPARATION FOR FILLET WELDS MIG WELDING (WELDS IN ALL POSITIONS, NO BACKING)

| Welding | Thickness (mm) | Preparation | Remarks |
|---|---|---|---|
| 2 sides alternately or simultaneously, automatic flat welding | t > 4 |   | |
| 1 side | t > 4 |   | If possible 1 back pass on other side, 5 mm groove (*) |
| 1 side | t > 6 |   | α = 70 °. Back-weld if possible |

*(*) Where a back-weld is advisable, it must be welded after gouging to the base of the first pass.* Table 50

*(*) Taken from standard NF 87-010* "Aluminium et alliages d'aluminium – Soudage – Préparation des bords" (Aluminium and aluminium alloys – Welding – Edge preparation).

## CHEMICAL COMPOSITION OF FILLER METALS (*)

| Alloy | Si | Fe | Cu | Mn | Mg | Cr | Zn | Ti |
|---|---|---|---|---|---|---|---|---|
| 4043A | 4,5<br>6,0 | 0,6 | 0,30 | 0,15 | 0,20 | | 0,10 | 0,15 |
| 4045 | 9,0<br>11,0 | 0,5 | 0,30 | 0,03 | 0,05 | | 0,10 | 0,20 |
| 4047A | 11,0<br>13,0 | 0,6 | 0,30 | 0,15 | 0,10 | | 0,20 | 0,15 |
| 5356 | 0,25 | 0,40 | 0,10 | 0,05<br>0,20 | 4,5<br>5,5 | 0,05<br>0,20 | 0,10 | 0,06<br>0,20 |
| 5183 | 0,40 | 0,40 | 0,10 | 0,50<br>1,0 | 4,3<br>5,2 | 0,05<br>0,25 | 0,25 | 0,15 |
| 5556A | 0,25 | 0,40 | 0,10 | 0,6<br>1,0 | 5,0<br>5,5 | 0,05<br>0,20 | 0,20 | 0,05<br>0,20 |
| 5556 (**) | 0,25 | 0,40 | 0,10 | 0,50<br>1,0 | 4,7<br>5,5 | 0,05<br>0,20 | 0,25 | 0,05<br>0,20 |

*(*) According to standard EN 573-3, Part 3: Aluminium and aluminium alloys – Chemical composition, except for the 5556.* Table 51

*(**) According to the Aluminum Association.*

## 9.1 Repair of defective welds

If inspection (X-ray, ultrasonic etc.) reveals unacceptable weld imperfections then the weld must be repaired.

On material under 4 mm thick, defective areas can be removed with a rotary tungsten carbide cutter mounted in a pneumatic drill. The axis of rotation of the cutter must be parallel to the axis of the weld so as to avoid incipient cracks.

For material over 4 mm thick, the defective areas should be removed with a pneumatic hammer fitted with a gouge (25).

The weld is then repaired by the same process (TIG or MIG) as was used to make the initial joint. Minor imperfections are nearly always repaired by TIG welding however, thickness allowing.

*(25) Carbon arc gouging is not advisable as it may introduce carbon into the weld seam.*

*(26) Cf. Chapter 10, Section 10-2.*

### CHOICE OF FILLER METALS AS A FUNCTION OF THE ALLOY COMBINATION

Each combination has three possible choices - indicated where the lines intersect - depending on the selected criterion: Optimum mechanical properties: top line – Optimum resistance to corrosion: middle line – Optimum weldability: bottom line

The filler metal indicated is: 4 : series 4xxx → 4043A, 4045, 4047A – 5 : series 5xxx → 5356, 5183, 5556A

| *Alloy A* | | | | | |
|---|---|---|---|---|---|
| Wrought<br>5000 Series<br>Mg < 3% | 5<br>5 (a)<br>4 - 5 (b) | | | | |
| Wrought<br>5000 Series<br>Mg > 3% (a) | 5<br>5<br>5 | 5<br>5<br>5 | | | |
| Wrought<br>6000 Series | 5 - 4<br>5<br>4 | 5 - 4<br>5<br>4 | 5 - 4<br>5<br>4 | | |
| Wrought<br>7000 Series<br>without copper | 5 - 4<br>5<br>4 | 5 - 4<br>5<br>4 | 5 - 4<br>5<br>4 | 5 - 4 | |
| Cast<br>Si > 7%<br>(c) | 4 (e)<br>4<br>4 | 5 - 4 (e)<br>5<br>4 | 4<br>4<br>4 | 4 | 4 (d)<br>4<br>4 |
| *Alloy B* | Wrought<br>5000 Series<br>Mg < 3% | Wrought<br>5000 Series<br>Mg > 3% | Wrought<br>6000 Series | Wrought<br>7000 Series<br>without copper | Cast<br>Si > 7%<br>(c) |

Table 52

*(a) 5000 series alloys with more than 3.5% Mg are sensitive to intergranular corrosion when exposed to temperatures over 65°C and when used in certain aggressive environments (26).*

*(b) 5000 series alloys with less than 3% Mg and 3000 series alloys that contain magnesium may be sensitive to hot cracking.*

*(c) The mechanical performance of the weld depends on the internal soundness of the castings. Gassed materials and injection mouldings are considered to be non-weldable.*

*(d) The percentage of silicon in the filler wire must be as near as possible to that in the casting.*

*(e) The welding of aluminium-silicon castings (40000 series) to 5000 series alloys should be avoided where possible as Mg2Si intermetallics form in the weldment and weaken the joint.*

## 9.2 Cleaning

Very fine black deposits of "soot" can often be seen sticking to the surface of the metal at the edge of the weld seam after MIG welding, especially when 5000 series semis are welded with 5356 alloy as the filler metal.

4043A filler wire leaves no deposits (except possibly at the start and finish of the weld) provided the welding equipment is set correctly.

This "soot" consists of particles of oxides (of aluminium and magnesium) caused by small amounts of filler metal vaporising in the arc, the temperature of the arc being higher than the boiling point of aluminium and magnesium. The vapour immediately condenses on cold parts of the sheet near to the weld.

These deposits only affect the appearance of the weld and have no impact on its mechanical properties or corrosion resistance.

This "soot" can be brushed off with a metal brush. This should be done as soon as possible after welding as it becomes much more difficult to remove if left for several hours.

**DISTORTION OF PARTS IN COMPRESSION**

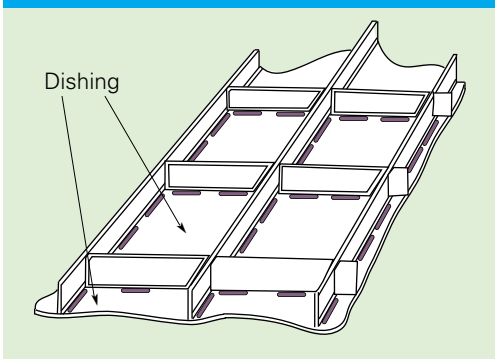



Figure 84

**EFFECT OF PRE-STRESS SHOT-PEENING**

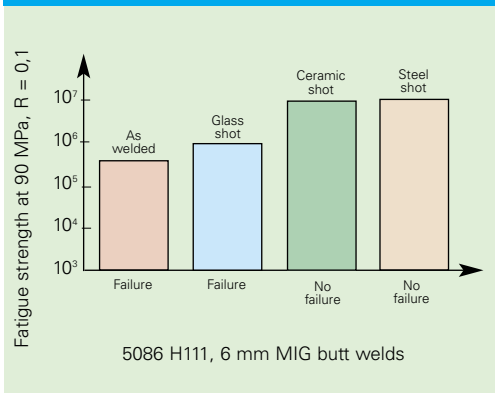



Figure 85

## 9.3 Correcting distortion

Minor distortion in sheet under 3 mm thick can be corrected with a hammer or mallet.

When sheets are bulged (figure 84), the welding torch can be used to apply "shrinkage heat" as locally as possible to the bulges. The heat makes these constrained areas expand (the welded zones are shorter than the sheet), and they are compressed. Rapid cooling – with a jet of water if necessary – then causes shrinkage which places the piece under stress and so corrects the warp. "Shrinkage heat" may also be combined with hammering.

It is trickier to apply "shrinkage heat " to aluminium than to steel because of the high diffusion of heat. Unlike steel, aluminium does not change colour so the temperature must be checked with tallow or thermocolour pencils.

Shrinkage heat does not affect the mechanical properties of 5000 alloys in the O or H111 condition. However it anneals 6000 series alloys and so reduces their mechanical properties.

## 9.4 Flush dressing of the weld

Flush dressing of the weld seam very significantly improves the fatigue resistance of the joint provided the seam is free from internal flaws which flush dressing would expose.

According to BS 8118 for example, shaving increases the endurance limit of a seam from 24 MPa for a 120° angle to 50 MPa for a dressed flush seam (27).

Welds are normally dressed flush with a fine abrasive wheel (50 to 80 grit).

## 9.5 Shot-peening

Shot-peening a weld seam puts its surface in compression, neutralising internal stresses detrimental to the weldment's fatigue strength.

Different types of shot can be used – glass, ceramic or steel – but it is the latter two which significantly enhance fatigue strength (figure 85).

Although there is no way of verifying the efficiency of these treatments, they can be applied to the welds of "hot spots".

*(27) Cf. figure 45, p. 65.*

# 10. INSPECTION

The purpose of inspection is to evaluate the quality of fabricated products and more specifically to grade the quality of a weld against an acceptable level of defects.

The acceptable level of defects is determined by a number of parameters:

- the load modes and load conditions – static and dynamic,
- the levels and variations of stress,
- the safety of persons and property,
- the technical and financial consequences of failure,
- the options for routine operational inspection and control.

## 10.1 Approval procedures

Approval procedures are contractual but they also make reference to standards (if any) and to the regulations of classification societies, especially as regards the qualification of welders.

They may be complemented by the fabricator's own inhouse procedures, governing welding methods in particular.

Tensile and bending tests are conducted on test specimens following approval procedures laid down by the classification societies. These tests are very important as they can help:

- to detect a lack of fusion that is hard to identify by NDT testing, and
- to adjust parameters so as to limit defects.

## 10.2 Testing welded joints

The frequency and extent of weld testing will depend on a number of criteria, such as:

- structure,
- rate of stress,
- any loads imposed on the welds.

In the course of fabrication it is possible to perform:

- non-destructive tests including random X-ray testing (28), ultrasonic etc.,
- visual inspection and dye-penetration (29) which can be performed over the whole of some welds to detect incipient cracks,
- tests of mechanical properties and bending tests on specimens taken from batches of welded metal according to the current methods

*(28) X-ray testing is not normally possible on fillet welds.*

*(29) According to NF A 09-120. Non-destructive tests. General principles of dye-penetration testing. June 1984.*

*(30) EN ISO 6520-1 Classification of geometric imperfections in metallic materials. Part 1: Fusion welding.*

# 11. WELD IMPERFECTIONS

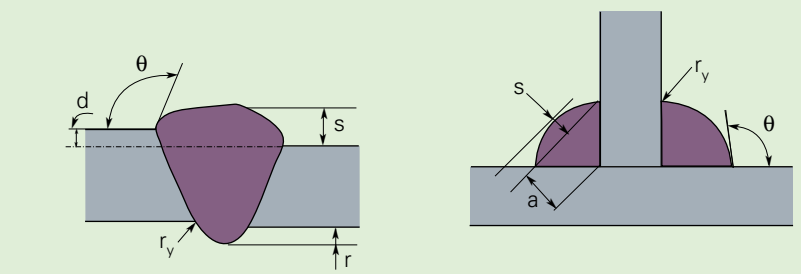
The causes of weld imperfections are numerous, and are a result of either the preparation of the metal or poor workmanship.

The most common defects encountered in aluminium welding are virtually the same as are found in the welding of steel: isolated cracks ('star cracks') or longitudinal cracks, incomplete penetration, poor bonding (fusion), porosity and undercuts.

Standards define weld imperfections based on measurements on a cross section (figure 86) of the weld and observations on its appearance.

An international nomenclature of defects has been established and is given in EN ISO 6520-1 (30) which lists 6 groups of imperfections, as shown in table 53, p. 104).

**GEOMETRICAL CHECKING OF WELDS**




Misalignment: d
Toe angles at base of bead: θ
Toe radii at base of weld: $r_y$

Toe angles at base of bead: θ
Toe radii at base of weld: $r_y$

*All numerical values are expressed in degrees or mm*

Figure 86

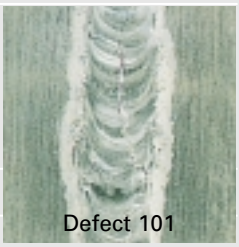
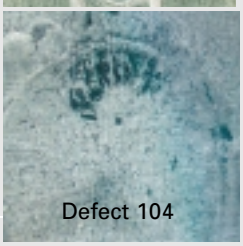
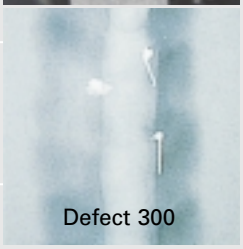
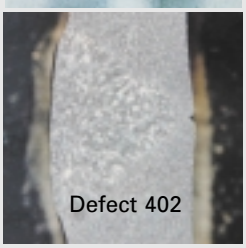
| GROUPS OF WELD IMPERFECTIONS | |
|---|---|
| Group | Type of Imperfection |
| 100 | Cracks |
| 200 | Cavities and wormholes |
| 300 | Solid inclusions |
| 400 | Lack of fusion and penetration |
| 500 | Defects of shape |
| 600 | Sundry defects |

Table 53

## 11.1 Common weld imperfections

Table 54 lists the most common imperfections together with their likely causes.

**TYPICAL WELD IMPERFECTIONS**

| N° | Type of Defect | Likely Cause | Photos of Imperfections |
|---|---|---|---|
| 101 | Cracks | Base alloy unsuitable<br>Poor choice of filler metal<br>Incorrect welding sequence<br>Excessive clamping<br>Sudden cooling |  Defect 101 |
| 104 | Crater cracks | Pass finished with sudden arc cutoff | |
| 2012 | Irregular wormholes | Work inadequately degreased<br>Work and/or filler wire dirty or wet<br>Insufficient protection by inert gas (low gas flow or leak in the system)<br>Pass begun on cold component<br>High arc voltage<br>Weld cooled too quickly |  Defect 104 |
| 2014 | Aligned wormholes | Incomplete penetration (double pass)<br>Temperature gradient between backing and work too abrupt<br>Excessive gap between edges of the joint |  Defect 2012 |
| 300 | Solid inclusions | Dirty metal (oxides, brush hairs) | |
| 303 | Oxide inclusions | Poor gas shielding<br>Metal stored in poor conditions<br>Castings | |
| 3041 | Tungsten inclusions (TIG) | Electrode diameter too small<br>Poor handling by welder<br>Excessive current density<br>Poor quality of tungsten electrode |  Defect 300 |
| 402 | Incomplete penetration | Inadequate cleaning (presence of oxide)<br>Incorrect bevel preparation on thick work (too tight, excessive shoulder)<br>Gap between workpieces too small (or incon-sistent)<br>Low current, especially at the start of the seam<br>Welding speed too fast<br>High arc voltage |  Defect 402 |

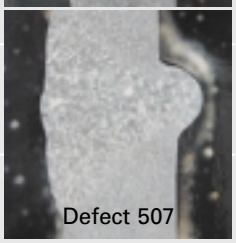
| N° | Type of Defect | Likely Cause | Photos of Imperfections |
|---|---|---|---|
| 4011 | Lack of fusion on edges | High arc voltage<br>Low current, especially at the start of the seam<br>Work cold (difference in thickness between materials to be welded) |  Defect 402 |
| 502 | Excessive thickness | Poor power control (poor U/I match)<br>Welding speed too slow<br>Poor edge preparation on thick work<br>Insufficient starting current |  Defect 502 |
| 507 | Misalignment | Work positioned incorrectly<br>Incorrect welding sequence | |
| 508 | Angle defect | Excessive welding power<br>Incorrect welding sequence | |
| 509 | Collapse | Wire speed too fast<br>Torch speed too slow<br>Poor torch guidance |  Defect 507 |
| 602 | Splatter (or beads) | Incorrect arc control<br>Problem in electrical contact to ground | |

Table 54

## 11.2 Effect of weld imperfections on fatigue strength

Some weld defects have a significant impact on the fatigue strength of the weldment:

- cracks (emergent or otherwise) and incomplete penetration are very serious flaws, as shown by tests carried out on weld defects [5] (figure 87),
- defects of geometry, especially sudden breaks in curves (angle at the base of the weld seam, misalignment etc.) aggravate stress intensity factors.

**BUTT WELD IMPERFECTIONS**

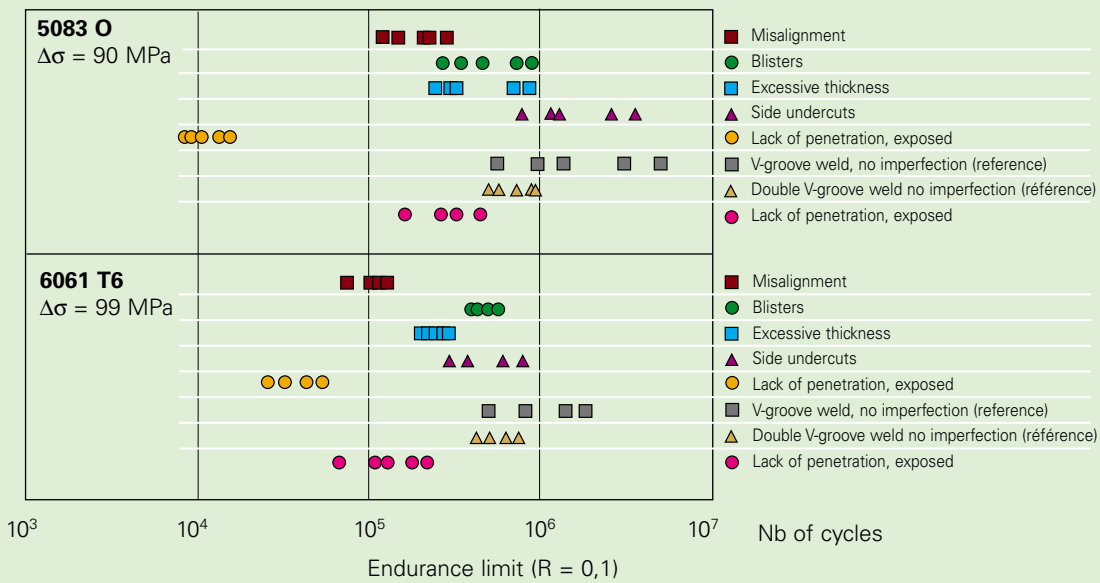



Figure 87

# 12. REPAIRS AND FITTINGS

European shipyards have responded to needs for the maintenance and modification of aluminium high speed ships by adapting to and specialising in this new activity [6, 7 8].

These yards repair damage to ships and modify onboard installations. The very long service life of aluminium ships means that from time to time they must be adapted in line with changing conditions of service, new equipment must be installed etc.

Work on aluminium alloy structures is based on classical sheet metalworking operations as is commonly carried out on steel ships (and their equipment), e.g. sheet and plate cutting, preparing edges for welding, making welds, correcting distortion etc.

The rules discussed previously for aluminium alloy forming and welding apply equally to these operations.

A number of basic precautions should be taken when welding items that are being repaired or modified:

- clean surfaces near to the weld with great care, using a brush to remove all traces of paint, oil or fuel that could have fouled the plates,
- dry thoroughly before welding to remove all traces of moisture,
- weld under cover of weather and away from draughts; if necessary, work under a tarpaulin when these operations are carried out in dock,
- pay particular attention to the direction in which welds are made – this will limit distortion and minimise the risks of hot cracking due to shrinkage,
- select the correct welding process: TIG (for work less than 6 mm thick) or MIG. TIG is more suitable for minor repairs where back access is difficult or impossible, being easier to use in such situations and providing better control of penetration than MIG.

For localised repairs such as a torn hull, the repair patch must be perfectly matched to the shape of the tear but will be bigger (achieved by hammering) to compensate for the contraction caused by welding. Without this precaution, the residual stress would attain a level where it would cause systematic cracking. The smaller the patch, the more pronounced this phenomenon.

Important note:
Never work with a torch or electric arc on or in any enclosed space, tank etc. that has held water (including seawater) or which has been in contact with moisture without first airing or thoroughly ventilating it to disperse the hydrogen produced by possible corrosion of the metal in contact with water. Failure to take this precaution may lead to an explosion hazard with consequences that could prove catastrophic for the operators (31). It is also a mandatory precaution for any work on fuel oil tanks.

# 13. LASER WELDING

Since the early Nineties, the uses of welding by laser (32) have spread widely in shipbuilding [9].

## 13.1 Principle of the laser

The laser is a device that generates an intense beam of coherent monochromatic radiation. In welding machines, this radiation is concentrated to obtain power densities in excess of $10^6$ W.cm$^{-2}$ which is sufficient for the industrial welding of aluminium alloys.

This power is used to generate a capillary filled with metallic vapour whose walls are lined with liquid metal in fusion. The resulting weld pool bath is displaced and the liquid metal solidifies after the beam has passed, ensuring metallurgical continuity between the workpieces (figure 88).

## 13.2 Welding lasers

Two types of industrial laser are used for welding metals:

- **in $CO_2$ lasers** the active medium is a gaseous blend of carbon dioxide ($CO_2$), nitrogen ($N_2$) and helium ($H_2$) at low pressure. The wavelength of the laser beam is 10.6 µm. Industrial $CO_2$ lasers can generate power ranging from 1.5 to 40 kW. The beam is transmitted by mirrors.

*(31) The amount of hydrogen that builds up in a ballast tank can be considerable even though corrosion is only superficial. In a tank with sides 1 metre long for example, i.e. 5 m² of area in contact with water, superficial corrosion one micron deep releases 16.8 litres of hydrogen !!!*

*(32) Light Amplification by Stimulated Emission of Radiation.*

■ **in Nd:YAG lasers** (Neodyme Yttrium Garnet), the active medium is a solid and the radiation wavelength is 1.06 µm, with a maximum available power of 3 to 4 kW. Despite their low power, Nd:YAG lasers offer a number of advantages over $CO_2$ lasers: the sources are more compact, and Nd:YAG beams can be carried by fibre optics which makes it possible to weld along complex paths using welding robots.

## 13.3 Laser welding of aluminium alloys

Aluminium alloys can be laser welded with no particular difficulty and at speeds as high as several metres per minute.

Laser welding offers a number of advantages:

- simplicity of preparation before welding,
- high welding speeds, several metres/minute on butt welds in 6 mm plate made from 5000 alloy,
- reduced distortion owing to the high welding speed and narrowness of the weldment,
- high penetration by the beam; it is possible to weld ($CO_2$ laser) 5000 series plate up to 12 mm thick in a single pass,
- high mechanical properties of the weld: nearly 90% of the parent metal on 5083 H116 and 70% for 6082 T6,
- different thicknesses can be welded,
- 'invisible' welding,
- good final condition (minimal finishing required),
- advanced automation.

Nevertheless laser welding requires close preparation tolerances and its energy efficiency is low.

## 13.4 Laser weldability of aluminium alloys

Aluminium alloys have a relatively low light absorption rate in the far-infrared: 3% with the $CO_2$ laser and 25% with the Nd:YAG laser. However this coefficient of absorption rises rapidly above fusion temperature and is approximately 90% when the material's vaporisation temperature is reached (figure 89).

For welding therefore, vaporisation of the metal must be initiated in the laser beam. Two very different types of interaction are observed according to the power density at the surface of the material (figure 90):

**COEFFICIENT OF REFLECTION OF LASER BEAMS**

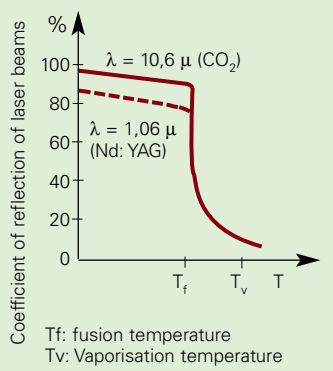



Figure 89

**LASER WELDING**

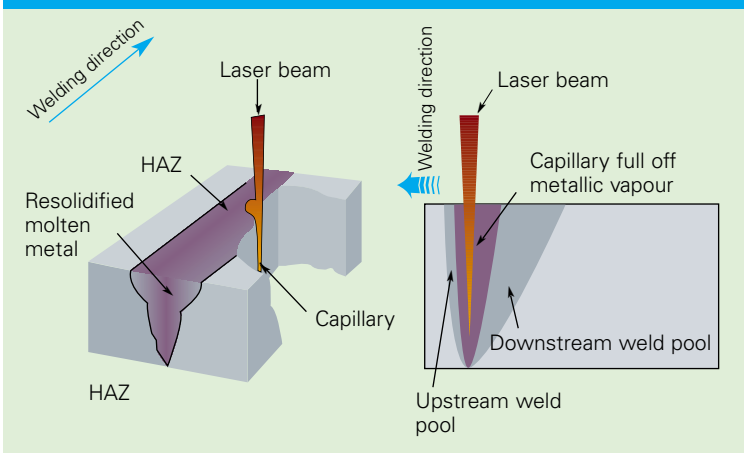



Figure 88

**INTERACTION BETWEEN LASER BEAM AND ALUMINIUM**

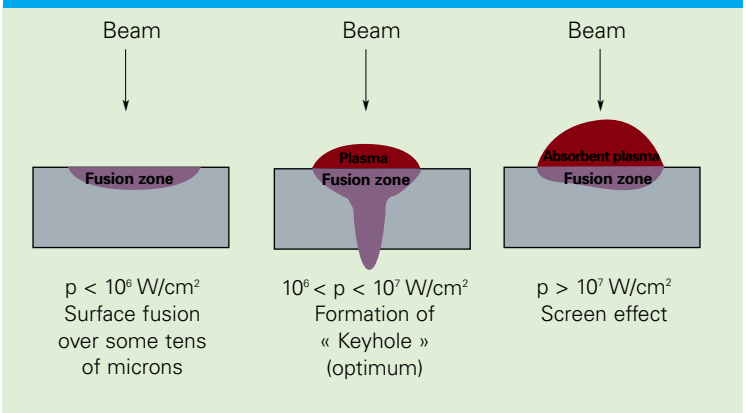



Figure 90

- at low densities, fusion is very superficial,
- at high densities a vapour capillary forms, i.e. a narrow and deep zone of fusion in the metal. It is this interaction which is needed for welding.

The threshold of interaction, i.e. the power density needed to form a vapour capillary, is of the order of $10^6$ $W.cm^{-2}$. The value of this threshold depends on the composition of the alloy – alloys that contain magnesium in the 5000 series (5754, 5083, 5086, etc.) have a lower threshold of interaction than other alloys (figure 91) and can be welded with less power.

It is important to note that using too high a power density is counter-productive as the metal vapours will form a plasma that acts as a shield. This is particularly true of $CO_2$ lasers.

A shielding gas must be used to prevent the immediate oxidation of the weld pool, and with $CO_2$ lasers the best results are obtained with argon/helium blends or pure helium. Argon can also be used with Nd:YAG lasers.

# 14. FRICTION STIR WELDING (FSW)

Friction welding with a tool (33) was invented by the TWI (34), the first patent being filed in December 1991 [10].

It is clear that this has been a decisive advance in the joining of metals in general and aluminium alloys in particular. In under ten years this new welding technique has enjoyed significant industrial development and growth in a number of sectors including shipbuilding, aerospace and the railways [11].

Since 1995 many publications have appeared and presentations given on the applications of FSW welding in shipbuilding at international conferences on High Speed Ships made from aluminium [12]. These publications reflect the obvious interest shown by naval architects and yards in this new technique, one which is already making very significant changes to aluminium shipbuilding and giving it fresh impetus [13, 14].

*(33) Friction stir welding (FSW).*

*(34) TWI: The Welding Institute.*

## 14.1 Principle of friction stir welding

The process is a simple one, consisting of shearing the metal without melting it (it turns 'pasty') with a rotating tool that has a 'probe' or pin on a level slightly below that of the weld. As it rotates the tool stirs the metal of the workpieces together and discharges it to the rear where the weld thus formed is softened and consolidated.

The metal is made to flow by the heat from the friction of the rotating shoulder against the surface of the metal. The shoulder, which is larger in diameter than the probe, contains the moving particles of metal and maintains a pressure that prevents the metal from being ejected outside the welded zone (figure 92).

The very significant forces that are exerted on the work mean that it must be clamped very firmly to the table of the welding machine.

## 14.2 Microstructure of the FSW joint

The specific properties of the FSW joint are due to its microstructure which is very different from the microstructure of an arc weld (MIG or TIG) owing to the simple fact that there is no process of fusion / solidification.
An FSW weld has four very distinct zones (figure 93) [15]:

- **zone A**, outside the weld, is the parent metal of each of the workpieces on either side of the joint. Its structure is unaffected by welding,
- **zone B** is the heat affected zone. It does not undergo any plastic deformation. As with the HAZ of conventional MIG or TIG welds, its mechanical properties

**THRESHOLD OF LASER INTERACTION**

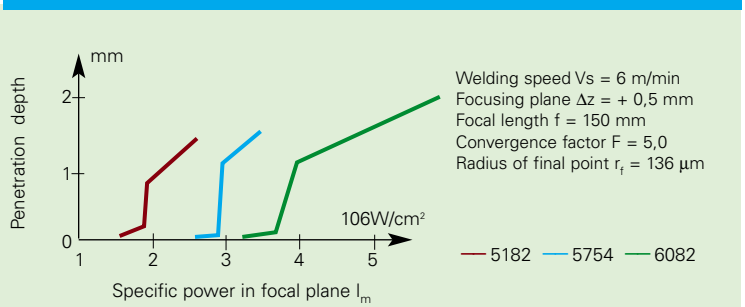



Figure 91

are low (figure 94). This zone is annealed in strain hardened alloys and over-aged in age hardened alloys (35). However no deformation occurs because the heating up of the metal and the temperature level attained are much lower than in arc welding,

- **zone C** is the thermomechanically affected zone that has undergone plastic deformation and heating. The structure of this zone depends on a number of parameters including the type of alloy,
- **zone D** is the "nugget" formed from recrystallised grains in which the metallurgical constituents of the parent alloys are dispersed. The grains are usually smaller than in the parent metal. This structure enhances the fatigue resistance of the welded joint.

In age hardened alloys the nugget is in a condition close to T4 (solution heat treated, natural ageing at ambient temperature) (figure 95).

## 14.3 Comparisons with arc welding

The FSW process operates at a temperature below the melting point of the metal, offering a number of advantages:

- **conditions of use** are simplified: surface preparation is confined to degreasing only. Where edge preparation is necessary, surfacing is adequate. The process requires no filler metal or shielding gas,
- **the applications** of FSW are far more extensive than with arc welding: all types of aluminium alloy products can be welded, whether castings or wrought semis,
- **the quality of the weld:** there are no risks of hot cracking (36) or porosity as hydrogen is not formed (37),
- **the quality of the assemblies:** distortion is minimal owing to the low temperature levels and the fact that welding takes place in a solid medium,

*(35) As a result the alloys are in the metallurgical condition indicated previously.*

*(36) It is possible to weld copper alloys (2000 and 7000 series).*

*(37) If hydrogen did form it would not be dissolved because its solubility in solid aluminium is zero.*

**FRICTION WELDING TOOLS**

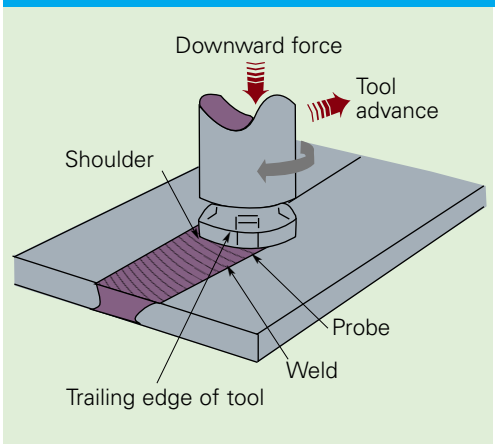
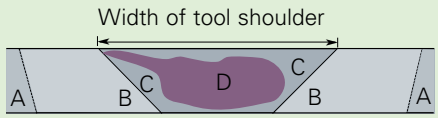



Figure 92

**MICROSTRUCTURE OF THE FSW JOINT**




A: Parent metal unaffected by weld
B: Heat affected zone (HAZ)
C: Unrecrystallised area found in aluminium alloys
D: Recrystallised nugget found in aluminium alloys

Figure 93

**CHANGE IN HARDNESS IN THE HAZ OF 5083 [15]**

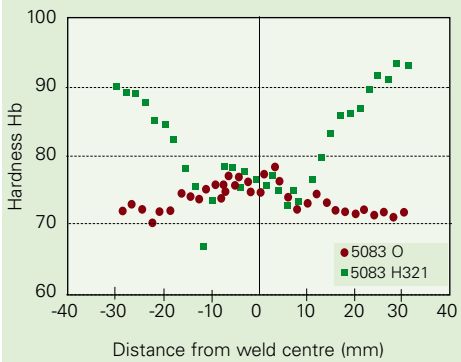



Figure 94

**CHANGE IN HARDNESS IN THE HAZ [16]**

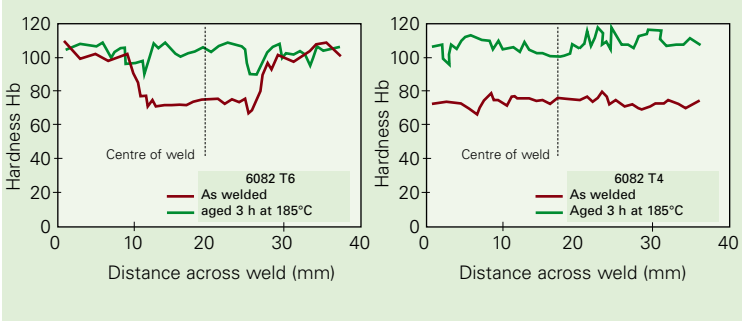



Figure 95

■ **environmental and working conditions:** there are no fumes, no flying particles of metal, no ozone emissions and no ultraviolet radiation. The process is also energy efficient, requiring about 20% of the power of MIG welding.

Its present state of industrial development makes FSW highly suitable for prefabricating sub-assemblies such as deck sections, walls, panels etc. [17] *in the workshop* for subsequent installation in ships and assembly by conventional welding techniques such as MIG (38).

A prototype "portable" machine designed by the University of Adelaide in Australia with The Welding Institute was presented recently [18]. This is in fact a tool connected to a hydraulic motor and mounted on a trolley for welding hull plates 5 mm thick. However although the tool is "portable", the components to be butt welded must be firmly fixed to withstand the forces necessary for welding.

*(38) A welding code is in the process of being approved by the classification societies.*

## 14.4 Possibilities of welding with FSW

In its present state of advance, FSW allows the welding of material up to 25 mm thick. Research into 6000 series alloys has shown that it is possible to go up to 50 mm thick with a single head (figure 96), and 75 mm with two heads (figure 97).

Given the current level of industrial development of the process, FSW can be envisaged in a number of configurations for butt welds and 'invisible' welds as shown in figure 98.

**WHORL™ TOOL**


Figure 96

**FSW WELDING WITH TWO HEADS**

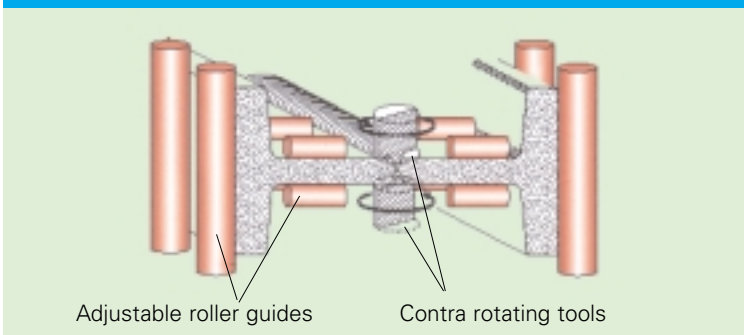



Figure 97

**TYPICAL FSW WELD CONFIGURATIONS**

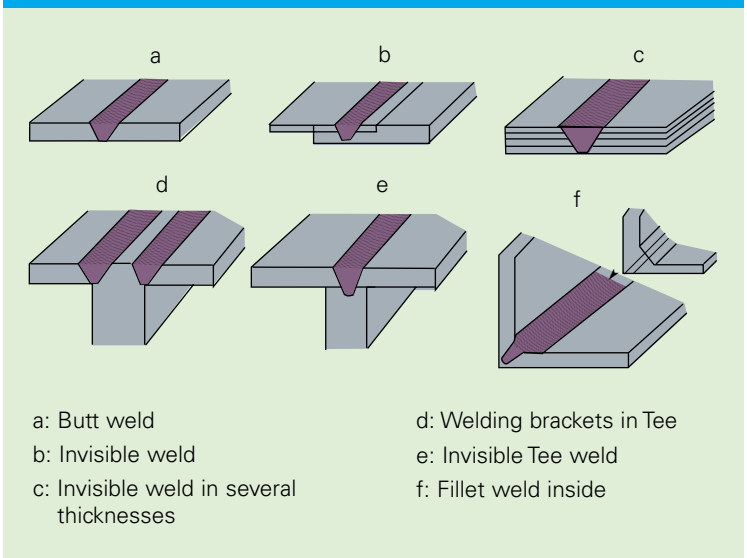



Figure 98

**MECHANICAL PROPERTIES OF WELDED 5083 AND 5383 6 MM THICK [20]**

| Alloy | Welding | $Rp_{0,2}$ (MPa) | Rm (MPa) | A % |
|---|---|---|---|---|
| 5083 H116 | MIG | 134 | 287 | 12,8 |
| | FSW | 157 | 335 | 17 |
| *Sealium®* | MIG | 150 | 308 | 13,5 |
| | FSW | 165 | 354 | 17 |

Table 55

**LIMIT OF ENDURANCE ON 5383 (AT $10^7$ CYCLES FOR R = 0.1) [11]**

| Alloy | Welding | Limit of Endurance (MPa) |
|---|---|---|
| *Sealium®* | Parent metal | 228 |
| | FSW | 172 |
| | MIG | 144 |

Table 56

**LIMIT OF ENDURANCE OF FSW JOINTS**

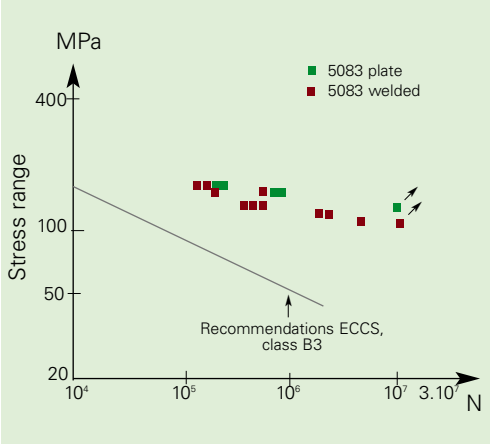



Figure 99

## 14.5 Performance of FSW welds

There have been numerous studies characterising the properties of FSW welds – their mechanical properties, fatigue strength and corrosion resistance of the weldment [19].

### Mechanical properties

The mechanical properties of FSW welded metal are superior to those of MIG welded metal (table 55).

Fractures usually occur at the edge of the friction zone, never inside it, most probably because of the strain hardening caused by the base of the tool.

The limit of elasticity is at least 10% higher in FSW welded metal than MIG welded.

### Fatigue resistance

The limit of endurance of FSW welded metal is superior to that of a MIG weld [21] (figure 99 and table 56).

The limit of endurance of an FSW weld is always superior to that of a MIG welded joint, and this is true for all alloys. This is because FSW ensures a very good connection between the joined workpieces. There is no 'sticking' (i.e. lack of fusion). It goes without saying that this applies only when the FSW joint is free from imperfections.

### Corrosion resistance

Investigations carried out so far have not indicated any particular sensitivity to corrosion by FSW welds. Their resistance to corrosion is at least equal to that of MIG or TIG welds.

ENGINE AND DRIVE SHAFT BEARER

# 15. STANDARDS

The main standards that govern the welding of aluminium are listed in table 57..

| MAIN EUROPEAN STANDARDS FOR WELDING OF ALUMINIUM | | |
|---|---|---|
| Reference | Date | Subject |
| BS EN 1011-4 | Dec 2000 | Welding. Recommendations for welding of metallic materials. Part 4: Arc welding of aluminium and aluminium alloys. |
| NF A 89-310 | April 1973 | Aluminium et alliages d'aluminium - Soudage - Assemblages élémentaires types - Critères de choix. |
| NF A 87-010 | April 1973 | Aluminium et alliages d'aluminium - Soudage - Préparation des bords. |
| BS EN 288-4/A1 | August 1997 | Specification and approval of welding procedures for metallic materials. Welding procedure tests for the arc welding of aluminium and its alloys. |
| NF A 89-220 | April 1973 | Aluminium et alliages d'aluminium - Soudage - Classification et contrôle des joints soudés. |
| BS 8118 | | Structural use of aluminium. Part 2. Specifications for materials, workmanship and protection. |
| BS EN ISO 9692-3 | Dec 2001 | Welding and allied processes. Recommendations for joint preparation. Part 3: Metal inert gas welding and tungsten inert gas welding of aluminium and its alloys. (ISO 9692-3:2000). |
| BS EN 12584 | June 1999 | Imperfections in oxyfuel flame cuts, laser beam cuts and plasma cuts. Terminology. |
| BS EN 30042<br>ISO 10042 | July 1994 | Arc-welded joints in aluminium and its weldable alloys. Guidance on quality levels for imperfections. |
| BS EN ISO 13919-2 | Dec 2001 | Welding -- Electron and laser beam welded joints -- Guidance on quality levels for imperfections -- Part 2: Aluminium and its weldable alloys (ISO 13919-2:2001). |
| BS EN ISO 6520-1 | Dec 1998 | Welding and allied processes -- Classification of geometric imperfections in metallic materials -- Part 1: Fusion welding (ISO 6520-1:1998). |
| NF EN 83-100-1 | Dec 1995 | Construction d'ensembles mécano soudés. Techniques de soudage. Partie 1 – Généralités: Terminologie, Classes de qualité de soudure – Etendue des contrôles. |
| BS EN 12062 | 1998 | Non-destructive examination of welds. General rules for metallic materials. |
| BS EN 970 | May 1997 | Non-destructive examination of fusion welds. Visual examination. |
| NF A 09-120 | June 1984 | Essais non destructifs. Principe généraux de l'examen par ressuage. |

| Reference | Date | Subject |
|---|---|---|
| BS EN 1289 | August 1998 | Non-destructive examination of welds. Penetrant testing of welds. Acceptance levels. |
| BS EN 1712 | Nov 1997 | Non-destructive examination of welds. Ultrasonic examination of welded joints. Acceptance levels. |
| BS EN 1713 | Sept 1998 | Non-destructive examination of welds. Ultrasonic examination. Characterization of indications in welds. |
| BS EN 1714 | Oct 1997 | Non-destructive examination of welded joints. Ultrasonic examination of welded joints. |
| BS EN 1712/A1 | February2003 | Non-destructive examination of welds. Ultrasonic examination of welded joints. Acceptance levels. |
| BS EN 1714/A1 | February 2003 | Non-destructive examination of welded joints. Ultrasonic examination of welded joints. |
| BS EN 444 | April 1994 | Non-destructive testing. General principles for radiographic examination of metallic materials by X- and gamma-rays. |
| BS EN 1435 | Oct 1997 | Non-destructive examination of welds. Radiographic examination of welded joints. |
| BS EN 12517 | Sept 1998 | Non-destructive examination of welds. Radiographic examination of welded joints. Acceptance levels. |
| BS EN 287-2 | June 1992 | Approval testing of welders for fusion welding. Aluminium and aluminium alloys. |
| BS EN 287-2/A1 | August 1997 | Approval testing of welders for fusion welding. Aluminium and aluminium alloys. |
| BS EN ISO 9956-10 | Nov 1996 | Specification and approval of welding procedures for metallic materials -- Part 10: Welding procedure specification for electron beam welding. |
| BS EN ISO 9956-11 | Nov 1996 | Specification and approval of welding procedures for metallic materials -- Part 11: Welding procedure specification for laser beam welding. |
| BS EN 12345 | June 1999 | Welding. Multilingual terms for welded joints with illustrations. |
| BS EN 1792 | 2003 | Welding. Multilingual list of terms for welding and related processes. |

Table 57

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